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Modelling of ductile fracture considering the effect of stress triaxiality and the energy partition theory in thin highstrength steel sheets

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Abstract It is well recognized in the literature that the fracture process of thin metal sheets involves three energy dissipation mechanisms i.e., plasticity, necking and surface separation. However, the complex stress state in thin structures hinders the experimental assessment of these quantities and, consequently, the failure modelling. This work evaluates the contribution of these mechanisms to the ductile damage of a thin advanced high strength steel sheet under different stress triaxiality ranges. The essential work of fracture test was carried out on a set of different notch geometry specimens that cover a wide range of stress states. The experimental trend of these specimens was simulated in ABAQUS/Explicit using a VUSDFLD subroutine. Bai and Wierzbicki uncoupled fracture model, which is a function of fracture plastic strain to stress triaxiality (η) and normalized Lode angle ($\overline{\theta}$), was selected as damage initiation criterion. A quantitative relationship of the fracture energy (G_0) as a

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Division of Mechanics of Solid Materials, Luleå University of Technology, 971 87 Luleå, Sweden function of (η) was proposed in this work and implemented in the model as a damage evolution law. The model captures well the experimental response and the influence of (η) on the softening behavior of the material. It was found that the sensitivity of G_0 to η is significant between 0.7 and 1.5. Above this rage, it seems that (η) has no influence on G_0 . The model showed also the relationship between the two local damage parameters (G_0) and the necking (G_n) with respect to the stress state. G_0 represents less than 10% of the total work of fracture, while the largest contribution comes from (G_n) .

 $\label{eq:complex-phase steel} \begin{array}{l} {\sf Keywords} \quad {\sf Ductile \ damage} \cdot {\sf Complex-phase \ steel} \cdot \\ {\sf Essential \ work \ of \ fracture \ \cdot \ Fracture \ energy} \cdot {\sf Stress \ triaxiality} \cdot {\sf FEA} \end{array}$

List of symbols

$\overline{\mathcal{E}}_P$	Equivalent plastic strain (dimensionless)
$\overline{\varepsilon}_{f}$	Equivalent fracture strain (dimensionless)
R	Notch radius (m)
η	Stress triaxiality (dimensionless)
θ	Lode angle (dimensionless)
$\overline{ heta}$	Normalized Lode angle (dimensionless)
$\bar{\sigma}$	Von Mises equivalent stress (Pa)
r	Third deviatoric stress invariant (Pa)
σ_m	Mean stress (Pa)
<i>σ</i> ₁ , <i>σ</i> ₂ ,	Principal stresses (Pa)
σ_3	
S	Deviatoric stress tensor (Pa)

D	Damage variable (dimensionless)
DI	Damage indicator for element deletion
	(dimensionless)
W_{f}	Total energy dissipated for fracture (J)
W _f	Total energy dissipated per unit area for
J	fracture (J/m^2)
W_p	Energy dissipated due to plasticity (J)
W_n	Energy per unit volume dissipated due to
r	plasticity (J/m ³)
W_e	Essential work of fracture (J)
We	Essential work of fracture per unit area (J/
	m ²)
W_n	Energy dissipated due to necking (J)
G_n	Energy per unit area dissipated due to
	necking (J/m ²)
W_O	Fracture energy dissipated due to material
	separation (J)
G_0	Fracture energy per unit area dissipated due
	to material separation (J/m^2)
l	Ligament length (m)
l_E	Characteristic length of a finite element (m)
t	Thickness of the test specimen (m)
h	Height of the localized plastic zone (m)
L	Length of the tensile specimen (m)
f_t	Tensile strength (Pa)
δ	Crack opening (m)
δ_{co}	Critical crack opening (m)
Α	Cross section (m^2)
a	Crack length (m)

1 Introduction

Thin-walled metallic structures are used in the automotive industry to manufacture high performance parts. High strength materials are commonly applied to achieve vehicle lightweighting, among which Advanced High Strength Steels (AHSS) are the dominant ones to manufacture safety-related components.

The manufacturing process of these components involves hot and cold sheet forming processes that require plastically deforming the material beyond the yield strength. One of the important mechanisms characterizing the failure of thin metal structures is tensile flow localization or necking followed by a ductile fracture. Therefore, several theoretical and numerical models have been developed over the years to predict accurately their failure while considering their characteristics, such as stable crack propagation prior to final fracture and large plastic zone ahead of the crack tip. Recently, the crack propagation resistance measured in the frame of fracture mechanics, i.e. the fracture toughness, has been postulated as a good descriptor of fracture resistance and crashworthiness (Frómet, et al. 2017).

Complex-phase (CP) steels belong to the first generation of AHSS family and show an excellent combination of strength and fracture toughness. Their yield strength and ultimate strength could range between 600 and 1000 MPa and between 800 and 1200 MPa, respectively, while showing high fracture toughness (more than 250 kJ/m²) and maintaining good ductility (7–15%) (Hilditch et al. 2015). Thanks to such outstanding properties combination, CP steels are mostly applied to manufacture parts requiring high fracture resistance, as components exposed to edge cracking during sheet manufacturing, or safety-related parts that require high energy absorption during crash events, such as sills, bumper brackets, impact beams etc., (Frómeta et al. 2017). Accordingly, the fracture resistance and damage mechanisms should be well understood to develop new steel grades, and to model the crash performance of thin-walled automotive parts.

Damage mechanisms controlling ductile fracture in structural metals at a micro-scale have been explored by several researchers in the past decade through micromechanical studies. It was found that microvoids nucleation, growth and coalescence is the major process controlling the local ductility and eventually leading to failure (Van Stone et al. 1985; Garrison and Moody 1987; Pardoen and Hutchinson 2000; Zhu et al. 2018; Koplik and Needlemen 1988; Gao et al. 2005). Stiffness degradation and strain softening results in progressive damage involving nucleation and growth of microscopic voids, prior to final rupture by voids coalescence. The flow curve can be obtained by uniaxial test up to diffuse necking due to uniform stress distribution at the cross-section. Beyond this point, the stress distribution is no longer uniform, and a high triaxiality stress state takes place.

Earlier studies by McClintock (1968) and Rice and Tracey (1969), on the role of the stress state on ductile fracture strain revealed the strong dependency of the growth of both long cylindrical voids and spherical voids on the hydrostatic pressure. Hancock and Mackenzie (1976) conducted experimental tests on notched round bar specimens covering a range of axisymmetric tension stress states. They reported that the strain to initiate fracture in a ductile solid correlates with the stress triaxiality. Hancock and Brown (1983) through plane strain and axisymmetric tensile specimens demonstrated that fracture strain is a decreasing function of triaxiality. Bao and Wierzbicki (2004, 2005) supported this argument through experimental and numerical results on different triaxiality regimes. However, they pointed out that the equivalent plastic strain to fracture is not a monotonic decaying function of triaxiality and presents some discontinuities, especially in low triaxialities.

More recent experimental and numerical studies showed that these models (accounting for triaxiality only) are not sufficient for the prediction of ductile fracture in low triaxiality and shear-dominated regimes (Xue 2007; Barsoum and Faleskog 2007; Kim et al. 2004). Later, it has been acknowledged that a new parameter needs to be considered to predict accurately the failure in ductile metals. This parameter is the Lode angle, which is related to the third invariant of the deviatoric stress tensor. Its influence can figure prominently in the coalescence of voids and stress carrying capacity of the material and, under constant triaxiality, the Lode angle is the dominant parameter controlling damage (Zhang et al. 2001; Gao and Kim 2006; Gao et al. 2009; Barsoum and Faleskog 2011; Yu 2015; Ma et al. 2017; Srivastava and Needleman 2013).

Bai and Wierzbicki (2008) suggested a new plasticity model to predict the ductility, accounting for the stress triaxiality and the lode angle simultaneously. Based on the classical plasticity theory, a 3D asymmetric failure surface considering the equivalent strain at fracture, stress triaxiality and Lode angle parameter was introduced. The calibration method of the fracture locus was also introduced and discussed in their work. Later, Bai and Wierzbicki (2009) proposed the Modified Mohr Coulomb criterion (MMC). Their model combined the Mohr-Coulomb model used in geomaterials with a new hardening model considering the pressure and Lode angle. The new model has been applied successfully in many studies to predict crack initiation and its direction (Dunand and Mohr 2011; Li et al. 2010; Sandin et al. 2021; Algarni 2015).

The energy dissipation mechanisms during a fracture process have been a research topic for many years (Cotterell and Reddel 1977; Hosford and Atkins 1990; Mai and Powell 1991; Tvergaard and Hutchinson 1992; Mai 1993; Pardoen et al. 2004). The total energy dissipated during a fracture process can be split into one part developed in the volume of the material ahead of the crack, due to plasticity, and another part spent in the fracture process zone (FPZ), due to damage and material separation. The Essential Work of Fracture (EWF) approach, first introduced by Cotterell and Reddel (1977), is one of the most effective and easy test methods used for the energy partition in metallic materials, following: $W_f = W_p + W_e$, where W_f is the total work, W_p is the plastic contribution, and W_e is the essential work of fracture. For a given thickness (t), W_{e} is proportional to the fracture area, so the ligament length (l), and W_p is proportional to the volume of the plastic zone, thus (l^2) . Therefore, the energy equation can be written as follows: $W_f = w_e t l + \beta w_p t l^2$, where w_p is the plastic work per unit volume of the plastic zone, w_e is the essential work per unit area of crack advance and β is a shape factor. In ductile fracture with extensive plastic deformation, W_e also is the contribution of two dissipation mechanisms: necking and material separation $(W_n + W_0)$, where W_n is thickness dependent (Pardoen et al. 2004).

Modelling of ductile fracture of AHSS has to consider the above-mentioned aspects to properly estimate the sheet fracture behavior during manufacturing and crash performance of structural automotive components. Accordingly, this work investigates the ductile fracture behavior of a new generation of CP steels with enhanced formability through an experimental-numerical study. Fracture mechanics tests under different stress states were conducted using notched specimens, with four notch radii values, including a pre-cracked condition, to cover different ranges of stress triaxiality. The MMC ductile fracture criterion and a quantitative relationship between the fracture energy and the triaxiality postulated in this work were used to model crack initiation and propagation. The material model was implemented in ABAQUS/Explicit using USDFLD subroutine. The predictive capability of the model, as well as a discussion on the energy partitioning, during the fracture process of this material, are presented herein.

2 Material characterization and experimental procedure

2.1 Material characterization

The studied AHSS is a new generation of CP steel, named CP1000HD, characterized by improved forming properties, high yield strength, high resistance to edge cracking and high crash energy absorption. The sheet thickness is 1.5 mm. The plastic strain- true stress behavior of CP1000HD (Fig. 1) was obtained using a Stiebler plastic hardening model (Stiebler et al. 1991) (Eq. 1). An inverse modelling approach was used to calibrate the materials parameters by fitting the force–elongation of a FE model of a tensile specimen (R3.75) to the experimental one. The calibration procedures are detailed in (Sandin et al. 2021).

$$\sigma(\varepsilon_p) = A + B\varepsilon_P + C(1 - \exp(-D\varepsilon_p)) \tag{1}$$

2.2 Stress state characterization and failure model calibration

The stress state in an isotropic solid material is conventionally defined by a 2nd order symmetric tensor [σ] and can be expressed in terms of three stress invariants (σ_m , $\overline{\sigma}$, r) or the three principal stress $\sigma_1 \ge \sigma_2 \ge \sigma_3$.

$$\sigma_m = \frac{1}{3} tr(\sigma) \tag{2}$$



Fig. 1 True stress-strain curves

$$\overline{\sigma} = \sqrt{\frac{3}{2}S:S} = \sqrt{\frac{1}{2}\left[(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2\right]}$$
(3)

$$r = \left[\frac{27}{2}\det([S])\right]^{1/3}$$
$$= \left[\frac{27}{2}(\sigma_1 - \sigma_m)(\sigma_2 - \sigma_m)(\sigma_3 - \sigma_m)\right]^{1/3}$$
(4)

where σ_m is the hydrostatic pressure, $[S] = [\sigma] - \sigma_m [I]$ is the deviatoric component of the stress tensor $[\sigma], \overline{\sigma}$ is the Von Mises stress, and r is the third invariant of the deviatoric stress tensor [S].

The prediction of ductile damage and fracture is receiving more and more attention in many sectors such as the automotive industry. However, the accurate prediction of failure especially in thin metal sheets, where high ductility is taking part, requires robust numerical models to account for all the material parameters involved in the failure mechanism. The MMC ductile fracture model, proposed by Bai and Wierzbicki (2009), has been successfully applied to predict ductile fracture initiation. The hydrostatic pressure (expressed in terms of stress triaxiality, η) and the third invariant of the deviatoric stress tensor related to the Lode angle parameter ($\overline{\theta}$) were used as stress state parameters for the model. The MMC fracture criterion represents a 3D fracture envelop that describes the monotonic decrease of ductility as a function of the increase of triaxiality (η) and the lode parameter $(\overline{\theta})$ as follow:

$$\varepsilon_{f}(\eta,\overline{\theta}) = \left\{ C2 \left[C3 + \frac{\sqrt{3}}{2 - \sqrt{3}} (C^{ax} - C_{3}) \left(\sec\left(\frac{\overline{\theta}\pi}{6}\right) - 1 \right) \right]. \\ \left[\sqrt{\frac{1 + C_{1}^{2}}{3}} \cos\left(\frac{\overline{\theta}\pi}{6}\right) + C_{1} \left(\eta + \frac{1}{3} \sin\left(\frac{\overline{\theta}\pi}{6}\right) \right) \right] \right\}^{-\frac{1}{C_{5}}}$$
(5)

where:

$$C^{ax} = \begin{cases} 1 for \overline{\theta} \ge 0\\ C_4 for \overline{\theta} < 0 \end{cases}$$

 η is the stress triaxiality, $\overline{\theta}$ is the normalized lode angle, and C_1 to C_5 are the shape parameters that have to be calibrated from experimental tests. The stress triaxiality (η) is the normalized pressure, defined as:

Table 1 Specimens designation and triaxiality range

Specimen	R15	Central hole	R3.75	Shear
η-range	0.4–0.5	0.33	0.5	0

$$\eta = \frac{\sigma_m}{\overline{\sigma}} \tag{6}$$

The Lode angle θ can be defined by the normalized third invariant (*r*) of the deviatoric stress tensor as follows:

$$\cos(3\theta) = \left(\frac{r}{\overline{\sigma}}\right)^3 \tag{7}$$

where θ is the lode angle $(0 \le \theta \le \frac{\pi}{3})$.

Therefore, the normalized Lode angle or the Lode angle parameter (-1 $\leq \overline{\theta} \leq 1$) is defined as:

$$\overline{\theta} = 1 - \frac{6}{\pi}\theta \tag{8}$$

For a proportional loading, the fracture surface is an asymmetric function of $(\overline{\theta})$.

In the study of Sandin et al. (2021), the stress state parameters $(\eta, \overline{\theta})$ for CP1000HD were calibrated through experimental tests on four flat tensile test specimens with different geometries that cover a large range of stress states (Table 1). The first two types of specimens are dogbone specimens designated R15 and R3.75 with a notch of 15 mm and 3.75 mm radii respectively and a gauge width of 7.5 mm. The flat central hole specimen has a hole of 7.5 mm diameter at the center with a gauge length of 15 mm and finally a pure-shear specimen. All the test coupons were machined perpendicular to the rolling direction of the metal sheet.

Digital Image Correlation (DIC) and stepwise modelling method by Marth et al. (2016) were used to obtain the equivalent fracture strain (ε_f) and the stress triaxiality (η) from the four tensile test specimens. With this data, it was possible to calibrate the MMC fracture locus for CP1000HD in Fig. 2 according to Bai and Wierzbicki (2009). The same curve was used as input for this study, with an extension to a high range of stress triaxiality up to 3 due to the high ductility of the material (Fig. 2).



Fig. 2 Modified Mohr–Coulomb fracture surface for the investigated materials. The red curve represents the plane stress state

2.3 Fracture toughness assessment

The fracture toughness of CP1000HD was assessed using the essential work of fracture (EWF) methodology. Double-edge notched tension (DENT) specimens with 200 mm long (L), 55 mm wide (W) and 1.5 mm thick (t) were machined perpendicular to the rolling direction of the same metal sheet (Fig. 3). The test was conducted on four batches of specimens with the same dimensions but different notch radii. Four notch radius sizes of 150 µm, 250 µm, 500 µm and 0.1 µm (fatigue pre-crack) were chosen to cover different ranges of stress triaxiality. The first three types of notched specimens were machined by electrical discharge machining. The fourth size i.e., a crack tip has a radius of 0.1 µm, was obtained by propagating a fatigue pre-crack of around 1 mm length from a notch root of 150 µm. The specimens herein will be designated as NR150, NR250 and NR500 for notch radii of 150, 250 and 500 µm respectively, and FPC for the fatigue pre-cracked specimens. A ligament length (l) varying between 6 and 14 mm was used for all the geometries, following the EWF procedure detailed in CWA 17793 (2021). For each ligament length, 2-3 specimens were tested to increase the precision of the experimental measurements. All the specimens were tested under a quasi-static loading rate of 1 mm/min. The displacement was measured directly on the surface of the specimens at an initial gauge length of 50 mm. An extensometer was used to measure the displacement on the surface of the



Fig. 3 Geometry of the test specimens

specimens during the loading from the elongation of an initial gauge length of 50 mm. A displacement control loading scheme was used to monitor the crack growth until the complete fracture of the test specimen.

Figure 4 shows the experimental force-displacement curves obtained from the EWF tests. It is obvious that all the test specimens of the same ligament length reached approximately the same maximum applied load, independently of the notch radius, except for FPC specimens. This difference in the maximum load capacity between FPC specimens and the other notch radii comes from the mismatch in the ligament lengths, as it is difficult to obtain the exact desired ligament length by fatigue pre-cracking. The forcedisplacement curves of NR150, NR250 and NR500 specimens display a long yield plateau, associated with the generation of a necking crack. The length of such plateau is higher for a larger notch radius. On the other hand, a shorter yield platform can be distinguished from the force displacement of FPC specimens, especially at large ligament lengths, because the high triaxiality at the crack-tip reduces the amount of necking that can be developed. Besides the length of the platform, the onset of damage and softening as well, significantly depends on the notch radius. The increase of the notch radius delays damage and fracture initiation. For instance, material softening and degradation initiates at approximately 0.25-0.4 mm for FPC, 0.4-0.5 mm for NR150 and NR250 and 0.55-0.7 mm for NR500, similar observation was also made by Omiya et al. (2022). After the onset of damage, it can be observed that FPC, NR150 and NR250 exhibit a monotonous stiffness degradation and softening until complete failure. Contrarily, NR500 shows a sudden drop in the load after a certain amount of stable crack advance, especially for larger ligament lengths.

The EWF results for the tested notch radii are plotted together in Fig. 5 Total work of fracture vs ligament length for DENT specimens with different notch radii. for comparison. As already reported by other authors (Vratnica 2010; Frómeta et al. 2019a; Pluvinage 2001; Kaufman 2001; Pluvinage and Gilgert 2003) the fracture toughness of the material is highly influenced by the notch radius. A $w_e = 304 \text{ kJ/}$ m² was obtained by pre-cracked DENT specimens i.e., the FPC ones. This is considered as the real fracture toughness of the material, while the results with notched specimens include also global plasticity and the obtained value cannot be considered as the material fracture toughness. Accordingly, as the notch radius increased, the fracture resistance increased resulting in values of 475 kJ/m², 514 kJ/m² and 606 kJ/m² for NR150, NR250 and NR500, respectively (see Table 2). It is also worth mentioning, that the plastic work, w_p , of all the tested geometries, except NR500 is nearly the same with $w_p = 33 \text{ MJ/m}^3$ for FPC, $w_p = 30 \text{ MJ/m}^3$ for NR150 and w_p equal to 32 MJ/m^3 for NR250, as denoted by the parallel regression lines. On the other hand, the linear regression of NR500 specimens yielded a lower plastic work value ($w_p = 23 \text{ MJ/m}^3$). Additionally, a discrepancy in the results of this notch radius can be clearly observed. The NR500 data points follow roughly a linear trend resulting in very low correlation coefficients ($R^2 = 0.59$) compared to the other geometries with R^2 equal to or above 0.9.

Pardoen et al. (2004) conducted a series of experimental tests on DENT specimens of different thin metal sheets of various thicknesses, aimed to separate



Fig. 4 Force–displacement curves of the DENT specimens with different notch radii and ligament lengths (*l*) of CP1000HD. **a** FPC; **b** NR150; **c** NR250; **d** NR500

the contribution of necking to the intrinsic energy related to crack tip fracture. According to his study, the energy dissipated in the fracture of thin metal sheets can be divided into three terms (Fig. 6). The first contribution comes from plasticity before necking. The plastic work scales with the crack advance and the size of the specimen. Therefore, W_p can be written as:

$$W_p = \int w_p dV \approx w_p LA \tag{9}$$

where *L* is proportional to the ligament length for the DENT geometry.

A typical response of a ductile metal subjected to uniaxial tension is represented in Fig. 6. Before the maximum load, the specimen goes through linear deformation, the straight-line region of the curve indicated by (A), followed by a plastic strain hardening (A-N). During this phase, the deformation is homogeneous and uniform all over the specimen.



Fig. 5 Total work of fracture vs ligament length for DENT specimens with different notch radii

 Table 2
 Values of each energy dissipation mechanism for different notch root radii in the DENT

	FPC	NR150	NR250	NR500
$w_p [\text{kJ/m}^3]$	33	30	32	23
$w_e [\text{kJ/m}^2]$	304	475	514	606
$G_n [\text{kJ/m}^2]$	294	445	474	556
$G_0 [\text{kJ/m}^2]$	10	30	40	50
G_0 / G_n	0.034	0.067	0.084	0.089

 w_e specific essential work of fracture, w_p specific plastic work, G_n fracture energy associated to necking, G_0 fracture energy associated to damage and material separation in the FPZ

The value of G_0 was obtained numerically

After the maximum load, necking initiates (N) and deformation starts to localize in an approximately constant wide h region, that increases with specimen thickness. Consequently, the specimen elongates in the loading direction and contracts in the thickness direction. Thus, the negative slope of engineering stress–strain curve. The energy dissipation associated with localized necking scales with the thickness of the specimen as follow:

$$W_n = \int w_n dV_L \approx h \overline{w}_n A \approx G_n A \tag{10}$$

where G_n is the fracture energy associated with necking per unit area of crack advance. As the load continues to increase the deformation localizes in a narrower region and strain reaches very high values, this point is defined by DI = 1 in Fig. 6 and, under



Fig. 6 Schematic illustration of the energy partitioning during ductile fracture process

monotonic loading, corresponds to the strain defined by Eq. 5. Within this region, ductile degradation starts by growth and coalescence of voids until final fracture. The energy associated with the growth and coalescence of voids in a unit cell is expressed as:

$$W_0 = G_0 A \tag{11}$$

where G_0 is the fracture energy per unit area of crack advance associated with the damage and material separation in the fracture process zone, and scales with the yield stress and the mean of voids spacing.

Accordingly, the energy dissipated in the fracture of ductile thin metal sheets can be divided into three terms:

$$W_f = W_p + W_n + W_0 = w_p L A + G_n A + G_0 A$$
 (12)

 W_p scales with the volume of the plastic zone, whereas W_n and W_0 scale with the cracked area, for instance, the ligament area in DENT specimens. In the EWF test, $G_0 + G_n$ are obtained by extrapolating the regression line to a ligament length equal to zero (Fig. 5) and designated as w_e and expressed in [kJ/m²]. As stated in (Pardoen et al. 2004), G_n is usually larger than G_0 , whereas w_p is the slope of the linear regression and is expressed in [MJ/m³]. The EWF equation is written as:

$$w_f = w_e + \beta w_p L$$
 where $w_e = G_n + G_0$ (13)

The detailed description of the energy partitioning during the ductile fracture, illustrated in Fig. 6, will provide the foundation for the experimental–numerical correlation, which will be discussed in detail in Sect. 4.

3 Implementation of ductile damage model

3.1 Finite element model

The numerical simulation was performed on the geometries considered in the experimental study using FE code ABAQUS/Explicit. A 3D solid element with reduced integration was used for all the models. Owing to the symmetry of the test specimens, only half of the specimens were modelled, and a symmetry condition was applied in the width direction following the x-axis (Fig. 7). A very fine and regular mesh was used in the ligament length zone where necking and fracture take place. The rest of the specimen was modelled with a coarse mesh. 8 elements were used through the thickness. To capture the stress singularity in the FPC specimens, the fatigue pre-cracks were modelled by removing a line of elements, that corresponds to the initial pre-crack lengths, from the mesh.

3.2 Constitutive model

The model is based on *J2* (Von Mises) yield surface with an associated flow rule, a nonlinear isotropic hardening is defined following Eq. 1 and Fig. 1. Large deformation was considered through "NLGEOM" function to account for the necking. Additionally, a smeared cohesive law is used to account for the energy due to damage.

3.2.1 Calibration of plasticity and ductile damage criterion

The true stress–strain curve in Fig. 1 was used as material input data for the model. The ductile damage criterion implemented in ABAQUS was used to reproduce the behavior of the material in this study. The onset of failure and progressive damage can be modelled by specifying a damage initiation criterion and damage evolution response. The ductile damage initiation criterion predicts the onset of damage due to nucleation, growth, and coalescence of microvoids in ductile metals and can be used in conjunction with Mises plasticity. The model assumes that the equivalent plastic strain at the onset of damage (ε_f) is a function of the stress triaxiality (η). The dependency on the Lode angle ($\overline{\theta}$) can also be introduced in the Explicit solver.



Fig. 7 Mesh and boundary conditions used for 3D finite element model of notched and fatigue pre-cracked specimens

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The damage evolution on the other hand defines the degradation behavior of the material once the initiation criterion has been met. The degradation of the material stiffness is associated with a scalar damage variable *DI*, defined as:

$$DI = \int \frac{d\varepsilon_P}{\varepsilon_f(\eta, \overline{\theta})} \le 1 \tag{14}$$

where $\varepsilon_f(\eta, \overline{\theta})$ is defined by Eq. 5 and the equivalent plastic strain (ε_p) is given by:

$$\varepsilon_P = \int_0^t \sqrt{\frac{2}{3}} \dot{\varepsilon}_{ij}^P \dot{\varepsilon}_{ij}^P dt \tag{15}$$

When the index DI = 1, ductile degradation starts by growth and coalescence of voids until the final fracture.

3.2.2 Modelling of fracture

A common way to model crack growth in an elasticplastic solid is the cohesive model, also known as traction-separation law. The concept was first proposed by Dugdale (1960); Barenblatt (1962) to describe the nonlinear fracture process in thin metal sheets. Dugdale (1960) assumes that within the plastic zone region, the stresses are constant and equal to the yield strength, while Barenblatt (1962) suggests a stress-strain softening relation that varies with the deformation. Hillerborg et al. (1976) were the first to apply the concept to model the cracking behavior of a concrete beam under bending. The model assumes that the material ahead of the crack tip is linear elastic until the stresses reach the tensile strength. The material starts then to soften and degrades but keeps its ability to transfer stresses across the discontinuity. The transferred stresses are a function of the crack opening (δ) and decrease linearly to zero at a critical opening (δ_{co}) , beyond which the crack propagates. The fracture process is governed by the amount of energy absorbed per unit area in opening the crack (G_0) , expressed as:

$$G_0 = \int_0^{\delta_{CO}} \sigma d\delta \tag{16}$$

The shape $\sigma(\delta_{co})$ curve has a great influence on the material response. The linear softening function in Fig. 6 is the simplest and most common shape. However, different shapes were also proposed such

as trapezoidal law used by Tvergaard and Hutchinson (1992) to predict toughness in ductile fracture due to void growth and coalescence.

A standard approach to implement cohesive law in commercial finite element codes is the crack band model of Bažant and Oh (1983). The concept is frequently used to predict failure in quasi-brittle materials and assumes that a prescribed localization band appears inside each element when a crack develops. The relation between the crack opening displacement and fracture strain is expressed through $\dot{\delta}_{CO} = \dot{\epsilon}_p l_E$, where l_E is the characteristic element length. The ductile damage model in Abaqus makes use of the crack band approach to reproduce the strain softening. After the onset of damage, the strain softening behavior of the material is characterized by a stress displacement response instead of a stressstrain. This requires introducing the characteristic length of the element (l_E) to the equation, thus approximating the width of the crack surface i.e., crack band width. This concept allows to supress the mesh sensitivity.

$$G_0 = \int_0^{\varepsilon_f} l_E \sigma d\varepsilon_p = G_0 = \int_0^{\delta_{CO}} \sigma d\delta$$
(17)

Tvergaard and Hutchinson (1992) also made use of the cohesive law in finite element calculation to study ductile failure under small-scale yielding. For mode I plane strain, the traction–separation constitutive model is defined by two independent parameters, the work of separation (G_0) and the tensile strength. The model introduces the cohesive length as a characteristic scale length and incorporates the effect of plastic strain in a modified traction separation relationship.

To account for the degradation of elastic stiffness, an internal scalar variable is defined and denoted herein by D. Physically, this internal damage variable represents the ratio of the area of the damaged material to the nominal cross-section area, therefore it takes values between 0 (for the undamaged material) and 1 (for the completely damaged material). D allows the transition of the elastic properties from the undamaged state to the damaged one according to (1-D) E. Its evolution according to ABAQUS (2012) User's Manual is as follow:

$$\dot{D} = \frac{\sigma_u}{2G_0}\dot{\delta} \tag{18}$$

where σ_u is the value of the yield stress when DI = 1 in Eq. 14. The value of *D* increases monotonously with the loading until reaching 1. Subsequently, the material loses its load-carrying capacity, and the element is removed from the mesh.

4 Results

4.1 Single G_0 value

The numerical model was initially fitted using a single value of fracture energy (G_0) as material input in ABAQUS. The FE force–displacement curves match well with the experimental ones for the different ligament lengths of the same geometry. However, a deviation from the test curves was observed when the same G_0 was used for the other notch radii. Therefore, different G_0 values were used to fit the different notch sizes as shown in Table 2.

In the study of Pardoen et al. (2004), the works of separation and necking were split by applying a linear regression to the total work of fracture as a function of the thickness for several materials. The ratio G_0/G_n obtained for 1 mm thickness sheets showed that both contributions are in the same order of magnitude for most of the tested material. For instance, a G_0/G_n of 1.1 was reported for a stainless steel. According to Siegmund and Brocks (2000), the separation energy represents 0.5% to 12% of the total dissipation energy in a fully plastic condition depending on the specimen geometry, size, and crack extension. It can be observed from in Table 2.

Table 2 that both the work of necking, G_n and the separation energy, G_0 in CP1000HD are not constant and both increase continuously with the increase of the notch radius. The largest contribution to the total work of fracture comes from the work of necking, whereas the work of fracture represents less than 10% of the total work of fracture in all the geometries. It is worth mentioning that the variation in the work of fracture seems to influence the work of necking. The increase of G_0 leaves room for progressive development of necking in the fracture process zone, leading to an increase in G_n as well. It results in an overall increase of the fracture toughness, in the measured w_e , as can be seen for DENT specimens with larger notch radius.

A similar trend was reported in previous studies. Frómeta et al. (2020) investigated the sensitivity of the EWF parameters to the notch radius in several AHSS steel grades, QP (quenching and partitioning steel), DP (dual-phase), TBF (Transformation Induced Plasticity (TRIP)-aided Bainitic Ferritic) and CP (complexphase). These steel grades showed different fracture resistance when tested with notched and fatigue precracked geometries. For instance, in CP w_e increased with the increase of the notch radius, meanwhile, w_p remained unaffected. On the other hand, for TBF and DP, we increased for notched specimens but w_p decreased for the fatigue ones. This seemed to affect smaller ligaments more than the larger ones. Similar for Q&P, but it seems to affect all ligaments. Similarly, Sunilkumar et al. (2021) explored the effect of notch radius on the fracture toughness in two automotive-grade dual phase steel (DP450) and IF (interstitial free) steel using EWF methodology. They came to the same conclusion that the sensitivity of the toughness to the notch radius is material dependent. The fracture energy sets the area under the softening curve. Any of the softening behavior represents an apparent change in the fracture energy, thus, the fracture toughness of the material. Accordingly, it is safe to say that the post-necking mechanical response of the material is also sensitive to the stress state.

CP1000HD belongs to this range of materials and that is the softening behavior after the maximum force in the force vs displacement curve cannot be described by a single fracture energy value. Therefore, to check the validity of this statement on this steel grade, a $G_0 - \eta$ relationship was defined and implemented as a damage evolution function in Abaqus.

4.2 G_0 dependency on the stress triaxiality

The work of fracture dependency on the stress state has been addressed by a few authors over the last decades. The influence of triaxiality on the cohesive parameters was studied by Siegmund and Brocks (1999, 2000). It was found that the increase of the stress triaxiality results in a nonlinear increase of the cohesive strength and a decrease of the cohesive energy. Accordingly, the effect of specimen geometry and size, as well as the amount of crack extension have to be considered. Pineau and Pardoen (2007) stated that this dependency is described by an exponential decay function for a range of triaxiality below 2–2.5. Above this range, the local fracture energy released ahead of the crack tip is independent of the triaxiality. Zhang et al. (2020) proved through experimental and numerical tests that the fracture energy is an exponentially decreasing function of the stress triaxiality.

In the present work, the triaxiality-dependent fracture energy function was implemented in ABA-QUS by means of a VUSDFLD user subroutine. The quantitative relationship between these two quantities is approximated solely by FE calculations, as obtaining a correlation function experimentally requires time-consuming and extensive fracture tests and FE computations. The values of G_0 in in Table 2.

Table 2 were used to roughly approximate the shape of the function in a triaxiality range between 0.5 and 3, then followed by an iterative method until the numerical force–displacement curve showed a good agreement with the experimental one. Figure 8 shows the variation of G_0 as a function of η obtained from these FE computations. The local work of fracture at the crack tip decreases rapidly and monotonously from 50 kJ/m² to 10 kJ/m² for a range of η below 1.6. This range corresponds to low-constraint geometries, as notched specimens. With the increase of triaxiality above 1.6 in the high constraint geometry i.e., FPC, G_0 reaches a plateau. The fracture behavior of the material in this range is described by constant G_0 values.

Figure 9 displays the fit between the FE models and the experimental ones. Only 3 ligament lengths for each geometry were plotted herein. Overall, a



Fig. 8 Correlation function between the local work of fracture G_0 and the stress triaxiality η obtained from the FE computations

satisfactory agreement between the numerical and experimental data was obtained. The subroutine has a good predictive ability of the material degradation rate and strain softening after the onset of damage in different stress states. The crack initiation in each model was detected by the first element deleted and was marked by an asterisk (*) on the force-displacement curves in Fig. 9. The stress constraints have a significant influence on the onset of the crack. In the high constraint geometry i.e., FPC, the crack initiates before the maximum tensile force. As the constraints ahead of the crack tip decrease with the increase of the notch radius, the onset of the crack is delayed. For the specimens with the lowest constraints i.e., NR250 and NR500, the onset of the fracture took place just before the softening and material degradation.

5 Discussion

Figure 10 shows the onset of the crack (DI = 1)indicated by an asterisk (*) on an exemplary forcedisplacement curve (l = 14 mm) for FPC, NR-150, NR-250 and NR-500. It can be seen that the initiation occurs at a different stage of deformation for each notch radius. The discontinous lines A, B, C and D on the Figure correspond to the displacements at which damage is activated (DI = 1) for the four geometries. It is evident that the crack initiation depends strongly on the notch size. As the notch radius decreases, DI reaches 1 at lower displacements. The initiation occurs largely before the peak load in FPC and NR150 (A and B), and around the maximum in NR250 (C). While, the crack initiates slightly before the final failure in NR500 (D). The distribution of the ductile failure parameters i.e., ε_P , η , $\overline{\theta}$, D and DI along the ligament length at different moments of the crack propagation corresponding A-D are displayed in Fig. 11. An increase in the plastic strain can be seen amid the increase of the notch radius, meanwhile the triaxiality decreased and the Lode angle increased in the positive range. Consequently, the ultimate strain increased, as shown in .

Figure. 2 In all the geometries except FPC, the first damaged element DI = 1 is at a certain distance from the crack front despite large ε_P . This is due to the large stress triaxiality in the middle part of the specimen. It can be presumed that the nucleation and growth of



Fig. 9 Experimental numerical validation of the Force-displacement curves using the triaxiality-dependent fracture energy function as damage evolution law. **a** FPC; **b** NR150;

c NR250; **d** NR500. The onset of the crack propagation determined numerically is marked in the plot by *

voids might have started inside the ligament zone and not at the notch root. One can also observe from Fig. 11 that the region of active damage 0 < D < 1 is restricted to a small area in front of the notch root and expands with the increase of the notch radius. The stress state close to the specimen edges is plane stress, while plane strain prevails in the center. Consequently, the crack grows at different stress conditions through the specimen thickness and propagates faster in the center, leading to a curved crack front also known as "thumb nail".



Fig. 10 a Numerical Force-displacement response for the exemplary selected ligaments: ($l \approx 13$ mm) for FPC and (l = 14 mm) for NR150, NR250, NR500; b damage index (*DI*) at failure plane for R500 at point D

Figure 12 displays the final failure of the four geometries using the proposed ductile damage model. According to the numerical computations, the low triaxiality in large notch radii specimens results in an increase of the fracture energy G_0 . Furthermore, the larger ε_P and necking ability that these specimens

exhibit compared to fatigue precracked ones, lead to an increase of G_n as well. Nonetheless, the damage variable (*DI*) reaches its maximum values at midthickness, in all geometries. As a result, the energy dissipated near the free edge is always larger than the center of the specimen.



Fig. 11 Evolution of variables at failure plane at different deformation stages A, B, C and D as defined in Fig. 10, with ε_p plastic energy, *DI* damage index, *D* damage variable, η stress triaxiality, $\overline{\theta}$ normalized Lode angle



Fig. 12 Contour plot of the damage index (DI) at a displacement of 0.765 mm (point D of Fig. 10) for the largest ligament length and different notch radii after fracture

6 Conclusion

This paper addresses the softening behavior at the post-necking stress-strain curve of a high strength CP steel sheet by considering the evolution of G_0 and G_n as a function of the stress triaxiality during crack initiation and propagation. The failure function was established numerically by means of a VUSDFLD user subroutine. Comparison of the numerically predicted force-displacement curves with the experimentally measured ones shows satisfactory results in terms of post-necking behavior for all four specimen types. This confirms the ability of the proposed failure function to successfully reproduce the softening of the material at different stress states. Then, the proposed approach can be used to address the fracture or damage modelling of thin-walled components or structures subjected to structural requirements.

The following conclusions can be drawn from the analysis of numerical models of CP1000HD with different notch radii:

- The softening behavior shown in the post-necking regime under different triaxiality conditions can be accurately estimated using an exponential decay function of the fracture energy (G_0) and the stress triaxiality (η).
- The local damage parameters (*G*₀ and *G*_n) are strongly dependent on the geometrical constraints, therefore the variability of the stress state should be considered in different constraint geometries.
- These two quantities are proportional, as G_0 increases with η , G_n increases too. However, this dependency is not linear.
- The sensitivity of the fracture energy G_0 to the stress triaxiality η is significant between

approximately 0.7 and 1.5. Above this range, η seems to have no influence on G_0 .

- The work of separation (G_0) represents less than 10% of the essential work of fracture (w_e) , with the largest contribution coming from necking (G_n) .
- The increase of the notch radius results in a decrease of the stress triaxiality, but an increase of the normalized Lode angle in the positive range and the ultimate failure strain (ε_f) .
- The damage variable reaches its critical value (*DI* = 1) at earlier deformation stage for low notch radius.

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Data availability No datasets were generated or analysed during the current study.

Declarations

Conflict of interest The authors have no conflicts of interest to declare that are relevant to the content of this article.

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