

#### Determining Fracture Properties for Predicting Damage Propagation from Notches in Composite Structures

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#### Introduction

- How can fracture mechanics be applied to predict failure of notched structures?
- Composites exhibit distributed damage rather than sharp cracks, but LEFM may still be applied
- FE of notched stiffened panel can give useful results
- What translaminar fracture properties to use?



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Xu et al, 2017

#### Overview

- Is translaminar fracture toughness a material or laminate property?
- Single value or R-curve?
- How to measure properties?







#### Translaminar fracture toughness

- Cannot measure Gc for a UD composite
- Material will split at the notch rather than fibre failure propagating from the crack tip
- Can back-calculate from laminate tests
- Splitting also occurs at notches in laminates
- Amount of damage and hence Gc depends strongly on the layup











# Effect of layup on Gc

• Increases with ply block thickness:

• Decreases with thin plies:

 Values typically higher for QI than CP due to greater damage



Layup

Li et al, 2009

[45/90/-45/0]25

 $[45_{2}/90_{2}/-45_{2}/0_{2}]_{s}$ 

 $[45_{4}/90_{4}/-45_{4}/0_{4}]_{s}$ 







Glc (kJ/m<sup>2</sup>)

104.2

212.3

297.2

## Effect of layup on Gc

• Damage zone size and hence Gc depends on layup



• Gc also depends on absolute thickness

• Gc is a laminate NOT a ply property



Li et al, 2009







#### Glc – single value or R-curve?

Some data suggests a single value of laminate Glc is sufficient, but other results indicate a significant R-curve for IM7/8552







#### Effect of layup on R-curve

QI laminates show greater R-curve than CP for same IM7/8552 Growth of larger and more complex damage zone



[90/45/0/-45]<sub>4s</sub> and [45/90/-45/0]<sub>4s</sub>







### Thin plies show less R-curve

- IM7/8552 t<sub>ply</sub>=0.125 mm [90/45/0/-45]<sub>4s</sub> OCT, 416x212mm
- Large damage zone, substantial R-curve
- MR40/K51, t<sub>ply</sub>=0.03 mm [45/90/-45/0]<sub>16s</sub> CT, 183x191 mm
- Very sharp crack, little R-curve











Sun, 2023





#### Measuring R-curves

- OCT reduces risk of compressive failure
- Tests on scaled OCT specimens
- 104x53, 208x106, 416x212 mm
- Linear increase in Glc continues as specimen size increases
- Even biggest specimen is too small!
- Increased risk of buckling







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### Validity of data reduction method

- · Glc may be invalid if crack is too long compared to ligament width
- E.g. ASTM D5045 criterion:

 $W-a > 2.5 (\frac{K_{IC}}{\sigma_u})^2$ 

- Tests on QI NCF material give invalid points
- Risk of misinterpreting R-curve if invalid points are not omitted
- Needs further investigation







### What is the crack length?

- Complex damage, not sharp crack
- In QI, 0° plies break before 45s
- "Crack" length is when all plies broken
- Region with broken 0s is process zone
- Include process zone in effective length
- Analogous to plastic zone in metals



#### Xu, Wisnom & Hallett, 2019





## Inconsistent results using initial crack length

• Four different specimens



104x53 300x210 300

300x210



 G curves plotted at the experimental damage initiation loads

• Not consistent – very different G







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### Consistent results when account for process zone

900x750mm



104x53, 300x210, 300x210,

G curves plotted at the initial damage

- propagation loads
- Based on failure of gauges at 5mm
- Now G values coalesce at 3mm
- Corresponds to fracture process zone size







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#### Measurement of fracture process zone

• OCT test interrupted just after peak



• Process zone of 3mm (red circle)



 Including process zone in effective crack length and G calculation gives consistent results for all four specimen sizes





### Conclusions

- Translaminar fracture toughness is a laminate not ply property
- Depends on layup, ply and absolute thickness
- R-curves also depend on layup
- Greater R-curve for QI and thick plies
- Need large specimens to measure full R-curve
- Consistent results by including fracture process zone in crack length and G calculation
- Enables LEFM prediction of structural response









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